

**B.TECH. (AEROSPACE ENGINEERING)  
(BTAE)**

**Term-End Examination**

**June, 2014**

**BAS-017 : FLIGHT MECHANICS**

*Time : 3 Hours*

*Maximum Marks : 70*

*Note : Q.No. 1 is compulsory. Answer any five questions from the remaining seven questions. Use of scientific calculator is permitted.*

1. Write short notes on the following : 8x2.5=20
  - (a) Induced Drag
  - (b) Service and Absolute Ceiling
  - (c) Neutral Point
  - (d) Dihedral Effect
  - (e) Static and Dynamic Stability
  - (f) Dutch Roll
  - (g) Weathercock Stability
  - (h) Critical Velocity (V-n diagram)
  
2. Draw the V-n diagram and explain it in detail. 10  
Also explain structural and aerodynamic boundary.
  
3. Discuss, in detail the determination of neutral and 10  
maneuver point from flight test.
  
4. Explain minimum drag speed and minimum 10  
power speed and their importance in aircraft  
performance analysis.

5. Explain the following in brief 5+5  
(a) Aileron reversal  
(b) Adverse yaw
6. An aircraft weighing 250000 N has a wing area 10  
of  $80 \text{ m}^2$  and its drag equation is  $C_D = 0.016 + 0.04 C_L^2$ . Calculate  
(a) Minimum thrust required ( $T_{\min}$ )  
(b) Minimum power required ( $P_{\min}$ ) for straight  
and level flight and the corresponding true air  
speeds ( $V_{\text{md}}$  and  $V_{\text{mp}}$ ) at sea level and at an  
altitude where  $(\sigma)^{1/2} = 0.58$ . Assume sea level air  
density to be  $1.226 \text{ kg/m}^3$ .
7. Discuss briefly the following : 5+5  
(a) Basic requirements of the rudder  
(b) Phugoid Motion
8. Explain, in detail the contribution of various 10  
components of the airplane to static directional  
stability.
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