

**B.Tech. AEROSPACE ENGINEERING
(BTAE)**

Term-End Examination

00423

December, 2016

BASE-003 : HIGH SPEED AERODYNAMICS

Time : 3 hours

Maximum Marks : 70

Note : Attempt any **seven** questions. All questions carry equal marks.

1. Derive the differential equations of motion for steady compressible supersonic flow. 10

2. (a) Define strong shock wave and weak shock wave in a compressible flow.
(b) Develop the famous Prandtl relation for normal shock waves. 3+7=10

3. Explain in detail the properties of hypersonic flow with neat sketches. 10

4. Discuss in detail 'Transonic Area Rule'. 10

5. Derive the Prandtl-Glauert equation using linearised subsonic compressible flow theory. 10

6. Deduce the following relation :

$$M_2^2 = \frac{1 + [(\gamma - 1)/2]M_1^2}{\gamma M_1^2 - (\gamma - 1)/2}$$

where M_1 and M_2 are values of Mach number ahead and behind the shock respectively.

$$\gamma = C_p \mid C_v.$$

10

7. What is hypersonic flow ? Discuss the viscous interaction effect on hypersonic flows.

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8. Explain with a neat sketch, the working and construction of a supersonic wind tunnel.

10

9. Discuss flow past forward swept wing. Also elaborate the effect of forward swept wing on stall characteristics and yaw motion.

10
