

**B.Tech. AEROSPACE ENGINEERING  
(BTAE)**

**Term-End Examination**

**00125 December, 2016**

**BAS-025 : SPACE DYNAMICS**

*Time : 3 hours*

*Maximum Marks : 70*

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*Note : Attempt any **seven** questions. All questions carry equal marks. Use of scientific calculator is permitted.*

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1. Sketch the velocity hodographs for elliptic, parabolic and hyperbolic motion and explain them in detail. 10
  
2. State Kepler's laws. Calculate the height at which a geostationary satellite should be positioned. 10
  
3. (a) Derive the equation of orbit for a spacecraft moving in the gravitational field of the Earth. 5  
  
(b) Explain the various approximate models of motion that are used to study the motion of a spacecraft. 5

4. Discuss the influence of ratio of injection and re-entry radius  $\rho_1$  and flight path angle  $\gamma_1$  on the angular range of a missile. 10
  5. What is the difference between Keplerian orbits and perturbed Keplerian orbits? 10
  6. Explain the difference between chemical rocket propulsion and electrical rocket propulsion systems used for spacecraft flights, with suitable diagrams. 10
  7. Explain the reference frame where the Sun is taken as origin and compare it with the reference frames usually considered for satellite orbits. 10
  8. Describe Cowell's method and Encke's method in detail. 10
  9. Explain the launch site and launch azimuth velocity penalty by making use of a plot. 10
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