

**B.Tech. AEROSPACE ENGINEERING  
(BTAE)**

**Term-End Examination**

00465

**December, 2014**

**BAS-012 : AERODYNAMICS – I**

*Time : 3 hours*

*Maximum Marks : 70*

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**Note :** *Attempt any **seven** questions. All questions carry equal marks. Use of scientific calculator is permitted.*

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1. (a) How do the laminar and turbulent boundary layers differ from each other ? Explain with neat sketches. 6
- (b) State and explain the Bernoulli's equation for incompressible flow. 4
2. (a) Draw the propagation of disturbance waves for subsonic and supersonic conditions. 4
- (b) Derive the fundamental equation for thin airfoil theory. State the assumptions made. 6
3. Explain downwash and induced drag. Also show how induced drag affects the angle of attack. 10
4. Derive an expression for vorticity in terms of the velocities in x, y and z directions respectively. 10

5. (a) Define angular velocity, strain rate and vorticity of a fluid element. 4
- (b) Consider the velocity field given by  $u = \frac{y}{x^2 + y^2}$  and  $v = \frac{-x}{x^2 + y^2}$ . Calculate the equation of the streamline passing through the point (5, 5). 6
6. Explain in detail the boundary layer separation with the help of neat sketches. 10
7. Explain the construction and working of a subsonic wind tunnel with the help of a neat diagram. 10
8. Define the following terms : 5×2=10
- (a) Wave drag
  - (b) Stall velocity
  - (c) Aerodynamic centre
  - (d) Neutral point
  - (e) Centre of pressure
9. Write short notes on any **two** of the following : 2×5=10
- (a) Drag Divergence Mach Number
  - (b) Newton's Law of Viscosity
  - (c) Fundamental Gas Laws
  - (d) Source and Sink Flow
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