

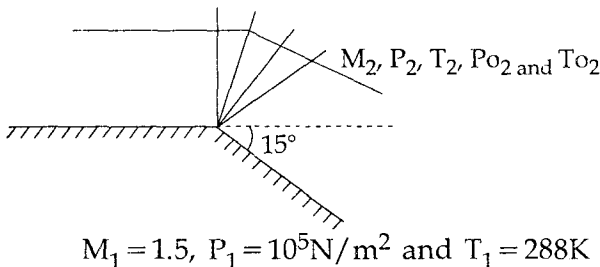
**B.Tech. AEROSPACE****Term-End Examination****December, 2013****BASE-003 : HIGH SPEED AERODYNAMICS**

Time : 3 hours

Maximum Marks : 70

- Note :** (i) Answer *any seven* questions.  
(ii) All questions carry *equal* marks.  
(iii) Use of scientific calculator is *permitted*.

1. Write a brief note on the reference temperature method and entropy layer effects on aerodynamics heating. 10
2. Explain the linearized theory for subsonic compressible flow about a thin wing at small angle of attack. 10
3. Discuss the performances of nozzles under various back pressures. 10
4. (a) A supersonic flow with  $M_1=1.5$ ,  $P_1=10^5\text{N/m}^2$  and  $T_1=288\text{K}$  is expanded around a sharp corner through a deflection angle of  $15^\circ$ . Calculate  $M_2$ ,  $P_2$ ,  $T_2$ ,  $P_{o2}$  and  $T_{o2}$ . 7



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|----|-----|--|----|
|    | (b) | Distinguish between Expansion Waves and Shock Waves.                                     | 3  |
| 5. | (a) | Derive the compressible flow Bernoulli's equation from one dimensional Euler's equation. | 7  |
|    | (b) | Define compressibility of a gas.   | 3  |
| 6. |     | Derive the Prandtl - Glavert relationship for two dimensional subsonic flow.             | 10 |
| 7. |     | Explain in detail shock polar with neat sketches.  | 10 |
| 8. |     | Write a brief note on "supersonic flows" with neat sketches.                             | 10 |
| 9. |     | Write a brief note on "Adiabatic steady state flow equations".                           | 10 |
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