

**B. TECH. (AEROSPACE ENGINEERING)
(BTAE)**

Term-End Examination

December, 2013

BAS-009 : INTRODUCTION TO AERONAUTICS

Time : 3 hours

Maximum Marks : 70

Note : Attempt *any seven* questions. All questions carry equal marks. Use of scientific calculator is *permitted*.

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| 1. | Define the following terms : | 10 |
| | (a) Angle of Attack | |
| | (b) Attitude of the Aircraft | |
| | (c) Stall | |
| | (d) Absolute ceiling | |
| | (e) Corner velocity | |
| 2. | (a) Define pitch, roll and yaw of an aircraft. Show with a proper sketch in Cartesian co-ordinate system | 5 |
| | (b) List down the various parts of an aircraft and describe them in brief. | 5 |
| 3. | (a) "Induced drag is a function of lift". Elaborate on the above statement. Show the effect of induced drag on an airfoil. | 5 |
| | (b) What is wake drag ? Explain with a suitable sketch. | 5 |
| 4. | (a) What are flaps ? Describe any two types of flaps. | 5 |
| | (b) Show that the relationship between velocity at minimum drag and velocity at minimum | 5 |

$$\text{power is } V_{P_{\min}} = \left(\frac{1}{3}\right)^{\frac{1}{4}} V_{D_{\min}}$$

5. Explain the following in brief : 10
- (a) TAS
 - (b) Static Margin
 - (c) Mach No.
 - (d) Advance Ratio
 - (e) Aspect Ratio
6. (a) Consider a straight wing of aspect ratio 6 with a NACA 2412 airfoil. Assuming low speed flow, calculate the lift co-efficient at an angle of attack of 6° . For this wing, the span effectiveness factor $e_1 = 0.95$, $C_{L_{\alpha_0}} = 0.105/\text{deg}$, $C_{L_0} = -2.2^\circ$ 5
- (b) What is the lift co-efficient for the same wing at the same angle of attack as in 6(a) above, but for a free stream Mach no. of 0.77 ? 5
7. Derive the expression for maximum climb angle for a jet propelled airplane. Also find the rate of climb corresponding to maximum climb angle. 10
8. (a) What is energy height ? Give the expression for the same and discuss its significance. 5
- (b) Consider an airplane with an instantaneous acceleration of 2.4 m/s^2 at an instantaneous velocity of 240 m/s. At the existing flight conditions, S.E.P. is 90 m/s. Calculate instantaneous maximum rate of climb that can be obtained at these accelerated flight conditions. 5
9. Describe the various types of engine used in an aircraft. Explain Turbojet and Ramjet engine, in detail. 10