No. of Printed Pages: 3

BASE-002

B.Tech. AEROSPACE ENGINEERING (BTAE)

Term-End Examination

00146

June, 2015

BASE-002: ROCKET PROPULSION

Time: 3 hours Maximum Marks: 70

Note:

- (i) Attempt any **seven** questions.
- (ii) All questions carry equal marks.
- (iii) Use of scientific calculator is permitted.
- (iv) Assume missing data suitably, if any.
- 1. Differentiate between any **two** of the following: 5+5=10
 - (i) Double base propellant and Composite propellant
 - (ii) Centre of pressure and Aerodynamic centre
 - (iii) Ram-Jet engine and Turbo-Jet engine
- 2. Explain the purpose and utility of rockets.

 Briefly describe the different sub-systems of a rocket.

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3. Is missile aerodynamics different from aircrafts' aerodynamics? Discuss each parameter that differentiates the two dynamics.

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- In a jet propulsion unit, air is drawn into the 4. rotary compressor at 15°C and 1.01 bar and delivered at 4.04 bar. The isentropic efficiency of compressor is 82% and the compression is uncooled. After delivery the air is heated at constant pressure until the temperature reaches 750°C. The air then passes through a turbine unit which drives the compressor only and has an isentropic efficiency of 78%, before passing the nozzle and expanding through atmospheric pressure of 1.01 bar with an efficiency of 88%. Neglecting any mass increase due to the weight of the fuel and assuming that R and y are unchanged by combustion, determine:
 - (i) The power required to drive the compressor,
 - (ii) The thrust per kg of air per second. 5+5=10
- **5.** Explain in brief the internal ballistic parameters, thrust coefficient and specific impulse. 5+5=10
- 6. Describe the working of Turbo-Jet engine. Also discuss its advantages and limitations.
- 7. Derive an expression for exit plane velocity for the flow through a rocket nozzle. Also explain why role of gamma (γ) is not observed on exit plane velocity.

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8. Explain in brief the future trends in rockets.

- 9. Write short notes on any two of the following: 5+5=10
 - (a) Guidance System of Missile
 - (b) Boost Sustained Trajectory
 - (c) Staging in Rockets