B.Tech. MECHANICAL ENGINEERING (BTMEVI)

Term-End Examination June, 2014

BIME-013: TURBO MACHINES

Time: 3 hours Maximum Marks: 70

Note: (i) Answer any seven questions.

(ii) All questions carry equal marks.

(iii) Use of scientific calculator is permitted.

- 1. (a) How does actual gas turbine cycle differ 5+5 from the theoretical cycle?
 - (b) What are the requirements of combustion chamber of a gas turbine?
- 2. What are the essential components of a gas turbine working on a closed cycle? Illustrate your answer with the help of a neat sketch. Derive an expression for the ideal thermal efficiency of such a plant.
- 3. Show by method of dimensional analysis that the resistance R to the motion of a sphere of diameter D moving with uniform velocity V through a fluid leaving density ρ and viscosity μ may be expressed

as
$$R = (\rho D^2 V^2) \phi \left(\frac{\mu}{\rho V D}\right)$$
.

- 4. A centrifugal pump has the following characteristics:
 outer diameter of impeller = 800 mm,
 width of impeller vanes at outlet = 100 mm,
 angle of impeller vanes at outlet = 40°,
 The impeller runs at 550 rpm and delivers
 0.98 cubic metres of water per second under an
 effective head of 35 m. A 500 kW motor is used
 to drive the pump. Determine the manometric,
 mechanical and overall efficiencies of the pump.
 Assume water enters the impeller vanes radially
- 5. A single stage single acting air compressor delivers 0.6 kg of air per minute at 6 bar. The temperature and pressure at the end of suction stroke are 30°C and 1 bar. The bore and stroke of the compressor are 100 mm and 150 mm respectively. The clearance is 3% of the swept volume. Assuming the index of compression and expansion to be 1.3, find:

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- (a) Volumetric efficiency of the compressor,
- (b) Power required if the mechanical efficiency is 85%, and
- (c) Speed of the compressor (rpm).
- 6. What is a centrifugal compressor? How does it differ from an axial flow compressor?
- 7. In a constant pressure open cycle gas turbine air enters at 1 bar an 20°C and leaves the compressor at 5 bar. Using the following data. Temperature of gases entering the turbine = 680°C, pressure loss in the combustion chamber = 0.1 bar, $\eta_{compressor} = 85\%$, $\gamma = 1.4$ and $C_p = 1.024$ kJ/kg K for air and gas, find :

at inlet.

- (a) The quantity of air circulation if the plant develops 1065 kW,
- (b) Heat supplied per kg of air circulation,
- (c) The thermal efficiency of the cycle.
- 8. In a gas turbine the compressor is driven by the high pressure turbine. The exhaust from the high pressure turbine goes to a free low pressure turbine which runs the load. The air flow rate is 20 kg/sec. and the minimum and maximum temperature are 300 K and 1000 K respectively. The compressor pressure ratio is 4. Calculate the pressure ratio of the low pressure turbine and the temperature of exhaust gases from the unit. The compressor and turbine are isentropic. C_p of air and exhaust gases = 1 kJ / kg K, and $\gamma = 1.4$.
- 9. (a) How are the steam turbines classified?

(b) Discuss the advantages of a steam turbine over the steam engines.

10. The velocity of steam exiting the nozzle of the impulse stage of a turbine is 400 m/s. The blades operate close to the maximum blading efficiency. The nozzle angle is 20°. Considering equiangular blades and neglecting blade friction, calculate for a steam flow of 0.6 kg/s, the diagram power and the diagram efficiency.

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5+5