No. of Printed Pages : 3

## **BAS-017**

## B.Tech. AEROSPACE ENGINEERING (BTAE)

## **Term-End Examination**

10473

December, 2018

## **BAS-017 : FLIGHT MECHANICS**

Time : 3 hours

Maximum Marks : 70

Note: Attempt seven questions in all. Question no. 1 is compulsory. Attempt any six questions from the remaining questions. Use of scientific calculator is permitted. Use given data for solving numerical problems.

Data for questions no. 2, 3, 4, 5 and 6.

Horizontal Tail Data Fuselage Data Wing Data  $C_{L_{0w}} = 0.21$  $C_{L_{\alpha_t}} = 4.5 \text{ per rad } C_{m_{0_f}} = -0.01$  $C_{L\alpha_{trr}} = 5.5 \text{ per rad}$   $S_t = 5 \text{ m}^2$  $C_{m_{\alpha_f}} = 0.1 \text{ per rad}$  $C_{m_{ac_w}} = -0.10$  $l_{+} = 5.5 \text{ m}$  $\tau = 0.5$  $b_{w} = 15 m$  $\eta_{t} = 0.9$  $C_{L_{trim}} = 0.3$  $\overline{C}_{--} = 1.75 \text{ m}$  $i_{+} = -1.5 \text{ deg}$  $C_{h_{\alpha_{+}}} = -0.015 \text{ per rad}$  $\lambda_{w} = Taper ratio = 1$  $C_{h_{\delta_e}} = -0.025 \text{ per rad}$  $X_{ac} = 0.25 \overline{C}_{u}$  $C_{L_{\delta_{\alpha}}} = 0.25 \text{ per rad}$  $X_{cor} = 0.30 \overline{C}_{ur}$  $i_{m} = 1.5 \text{ deg}$ e = Oswald's efficiency = 0.9**BAS-017** P.T.O

- 1. Distinguish between the following using sketches: 4+3+3
  - (a) Static stability and Dynamic stability
  - (b) Floating and Restoring characteristics
  - (c) Primary and Secondary control surfaces
- 2. Calculate (i) pitching moment coefficient at zero lift ( $C_{m_0}$ ), and (ii) pitching moment curve slope ( $C_{m_{\alpha}}$ ) for complete aircraft for stick-fixed case using given data. 5+5
- 3. Calculate (i) pitching moment coefficient at zero lift  $(C'_{m_0})$ , and (ii) pitching moment curve slope  $(C'_{m_{\alpha}})$  for complete aircraft for stick-free case using given data. 5+5
- Define neutral point. Calculate neutral points for stick-fixed and stick-free cases using given data. 2+4+4
- Define elevator control power. Derive expression for elevator control power and calculate its value using given data.
- 6. Define the elevator angle to trim and calculate its value using given data. 2+4+4

2

Take  $C_{L_{\alpha_w}} \cong C_{L_{\alpha}}$ .

**BAS-017** 

- 7. Explain the following using sketches (if required): 3+3+2+2
  (a) Dihedral effect
  - (b) Weathercock stability
  - (c) Adverse yaw
  - (d) Trim tab
- 8. Write notes on the following :
  - (a) In-flight measurement of stick-fixed maneuver point
  - (b) Methods of aerodynamic balancing
- **9.** (a) Discuss the cross-coupling of lateral and directional effects.
  - (b) Define rudder power. Explain the uses of a rudder. 2+3
- 10. Write short notes on any *two* of the following:  $5 \times 2$ 
  - (a) Power Effects
  - (b) Aerodynamics Balancing
  - (c) Phugoid

BAS-017

3

5 + 5

5