

**B.Tech. AEROSPACE ENGINEERING  
(BTAE)**

**Term-End Examination**

**December, 2015**

**BASE-002 : ROCKET PROPULSION**

*Time : 3 hours*

*Maximum Marks : 70*

**Note :**

- (i) *Attempt any seven questions.*
- (ii) *All questions carry equal marks.*
- (iii) *Use of scientific calculator is permitted.*
- (iv) *Assume suitable data, if missing.*

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1. Explain in detail the safety criteria of a missile launch for parent aircraft. 10
  2. Derive an expression for the time of climb of a missile. Explain the assumptions in detail. 10
  3. Discuss the basic inputs for design of a solid propellant rocket. 10

4. A turbo jet engine travels at 216 m/s in air at 0.78 bar and  $-7.2^{\circ}\text{C}$ . Air first enters the diffuser (of 0.78 bar) in which it is brought to rest relative to the unit and it is then compressed in a compressor through a pressure ratio of 5.8 and fed to a turbine at  $110^{\circ}\text{C}$ . The gases expand through the turbine and then through the nozzle to atmospheric pressure (0.78 bar). The efficiencies of diffuser, nozzle and compressor are each 90%. The efficiency of turbine is 80%. Pressure drop in combustion chamber is 0.168 bar.

Determine :

- (i) Air fuel ratio, and  
(ii) Total thrust, if the inlet cross-section of diffuser is  $0.12\text{ m}^2$ . 10

Assume calorific value of fuel as 44150 kJ/kg of fuel.

5. Describe the working of a Ram-Jet engine. Depict the various processes occurring in it on the h-s diagram. 10
6. Explain with a neat sketch all forces and moments acting on the two separating stages in a multi-stage rocket. 10
7. What is thermal protection system ? Describe a thermal protection system based on heat dissipation. 10

8. Differentiate between any *two* of the following :

$2 \times 5 = 10$

- (a) Mass loading and Volume loading concepts
- (b) Missile and Rocket
- (c) Homing command guidance and Beam rider guidance

9. Write short notes on any *two* of the following :

$2 \times 5 = 10$

- (a) Flight Dispersion
  - (b) Geysering
  - (c) Solid Propellants
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